

# ASPECTS OF USE OF CFD FOR UAV CONFIGURATION DESIGN

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# UCAV DESIGN PROBLEM

- Problem (for Aerodynamics) is as much due to novel planforms as Unmanned
- Novel planforms negate traditional Aerodynamic ground rules (sweep, span, AR etc)
- S&C is a significant challenge
- Requirement for rapid proto-typing for planform/basic layout studies and control surface optimisation
- Fast-response WT - small scale, stereo-lithography, PSP
- Fast-response CFD -Euler, High RE turb models RANS

# TYPICAL EXAMPLE

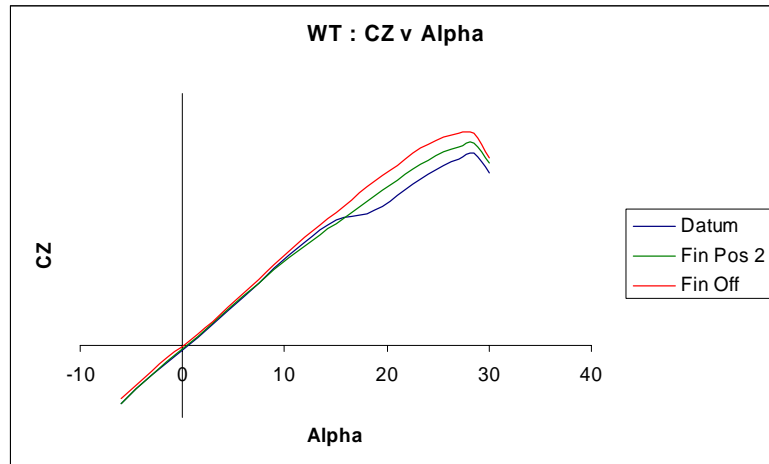
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## Investigation of Fin Position on a typical Novel Planform

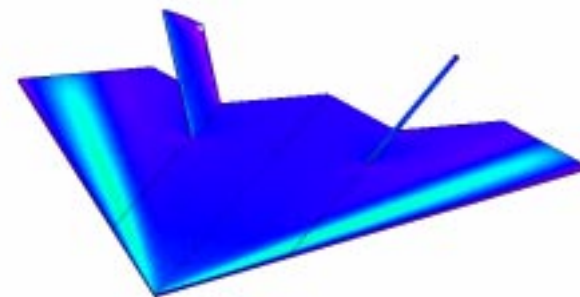
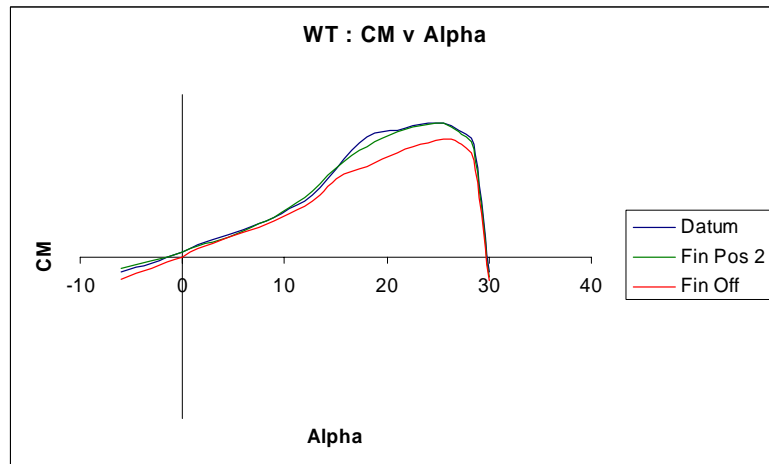
- Establish credibility of CFD for prediction of general flow trends at low speed, high incidence for novel planforms
- Assist in interpretation of 'small-scale' wind tunnel testing

# TYPICAL EXAMPLE

BAE SYSTEMS



Small Scale WT testing -  
Effect of fin position



Datum (flat-plate) model

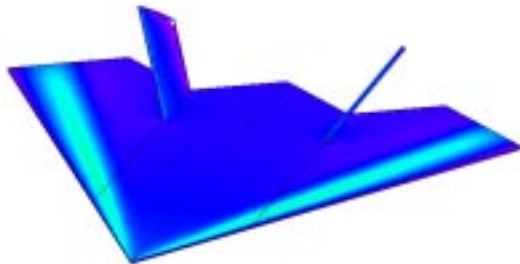
# NOTES ON CFD CALCULATIONS

- 6-8million unstructured grid cells required for credible vortex capture from Euler, with particular emphasis on field resolution
- 2-3million 'BAE Systems Autogrid' cells required for equivalent capture from RANS
- $k\epsilon$ RNG turbulence model (wall function) suitable
- Euler solution turnaround 4hrs on 8 Origin processors, RANS 2 days

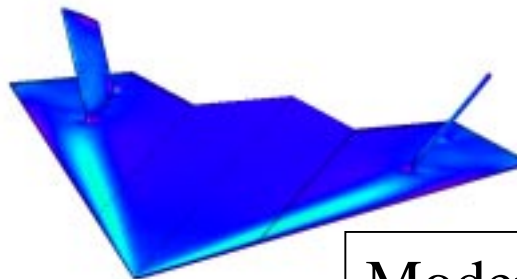
# EFFECT OF FIN POSITION

BAE SYSTEMS

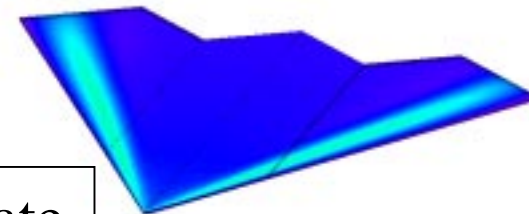
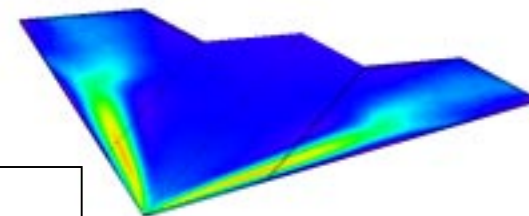
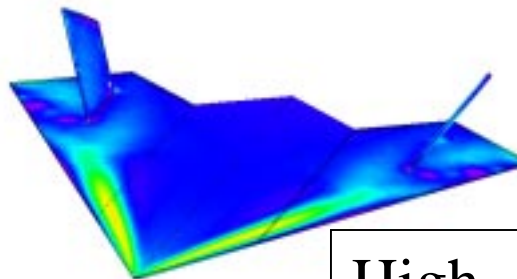
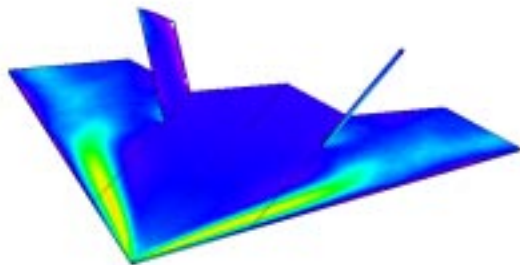
Datum



Fin Pos2



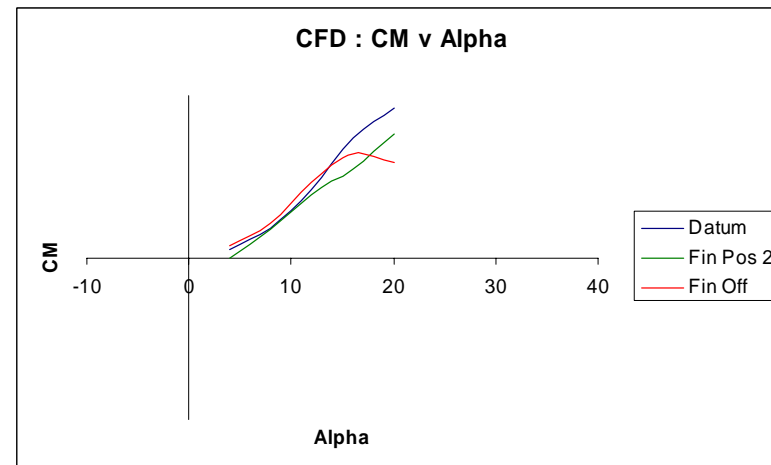
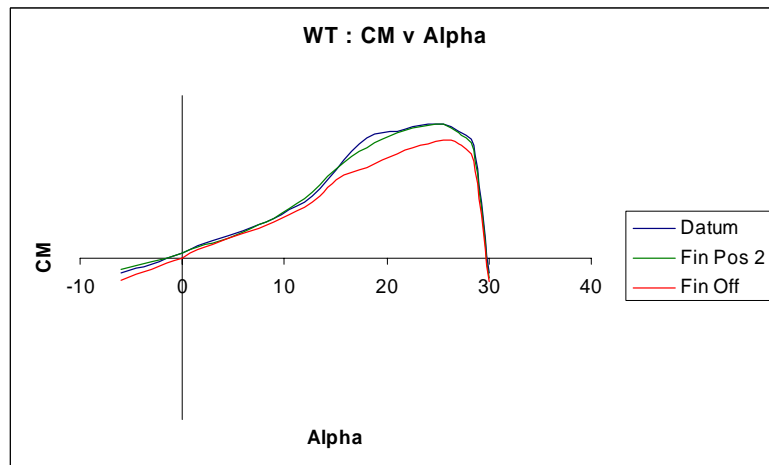
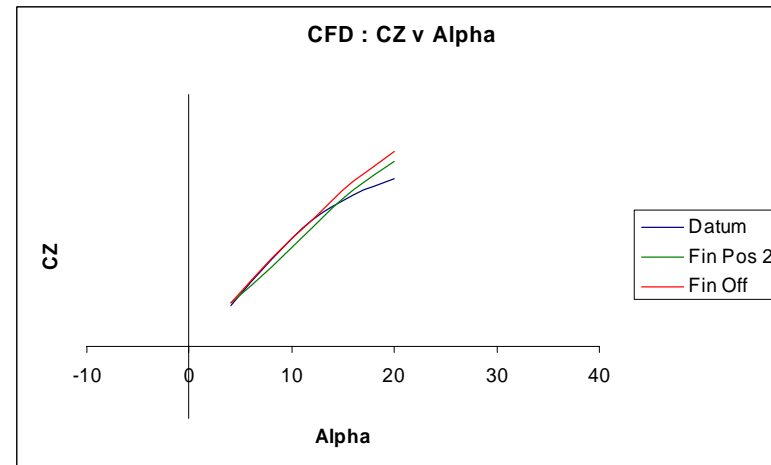
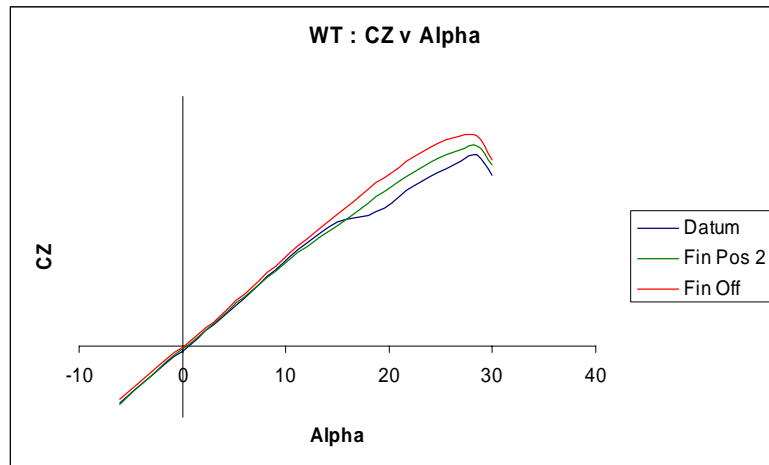
Fin Off

Moderate  
IncidenceHigh  
Incidence

Flat Plate CFD Euler, local velocity contours

# EFFECT OF FIN ON FORCES

BAE SYSTEMS

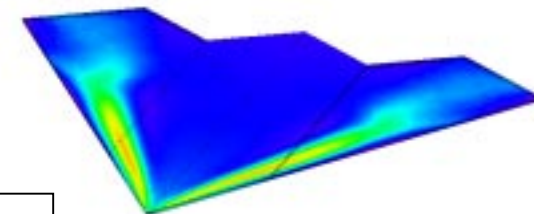
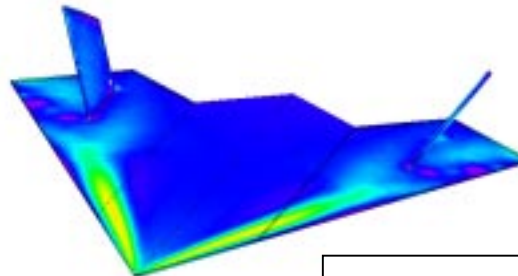
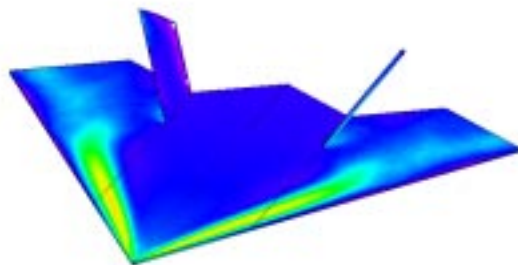
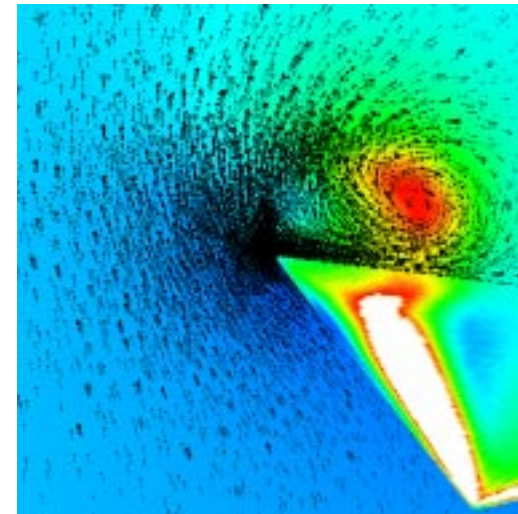
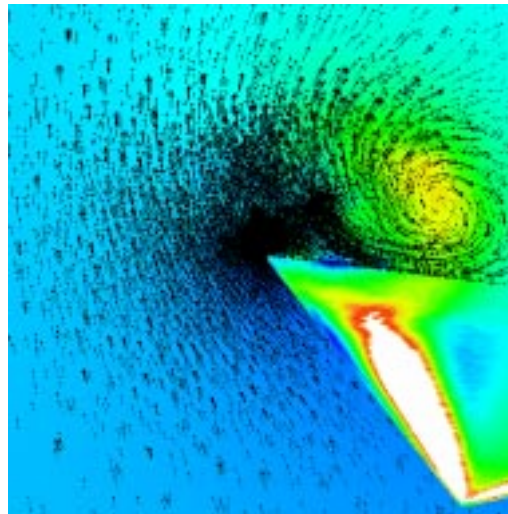
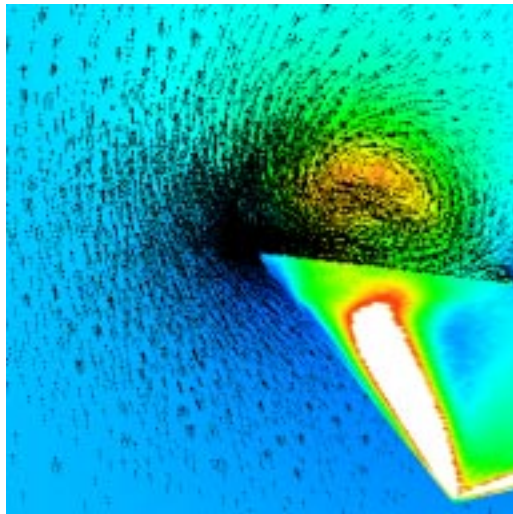


Flat Plate Wind Tunnel v CFD (Euler)



# EFFECT OF FIN ON FLOWFIELD

BAE SYSTEMS



High Inc

Datum

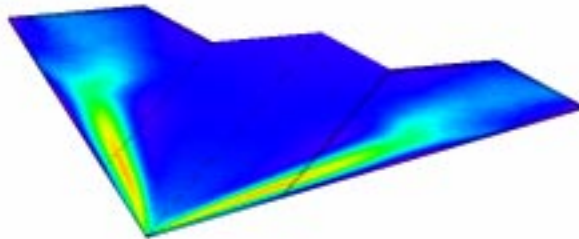
Fin Pos2

Fin Off

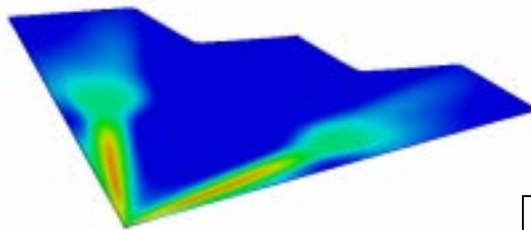
Flat Plate CFD Euler, velocity vectors, local vel contours

# INVISCID v VISCOUS

BAE SYSTEMS



EULER



RANS

High Inc

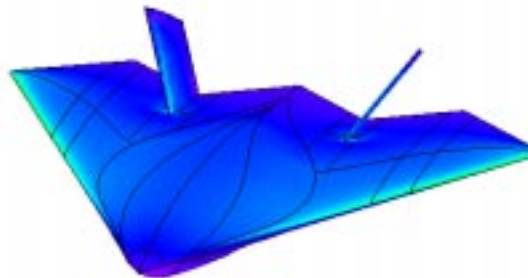
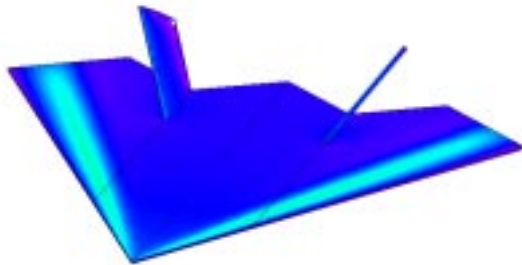
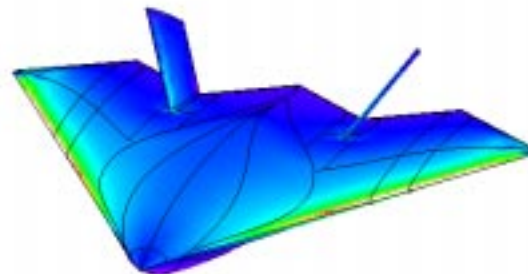
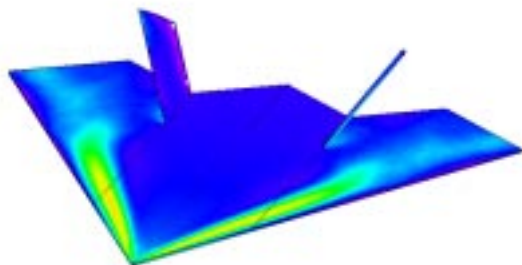
Flat Plate CFD Euler v RANS

# EFFECT OF THICKNESS

BAE SYSTEMS

Flat Plate

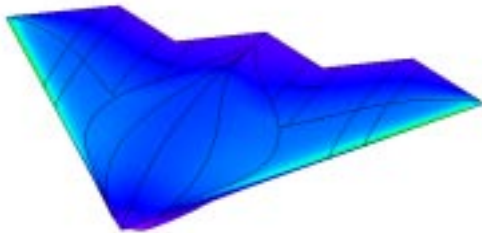
10% t/c

Moderate  
IncidenceHigh  
Incidence

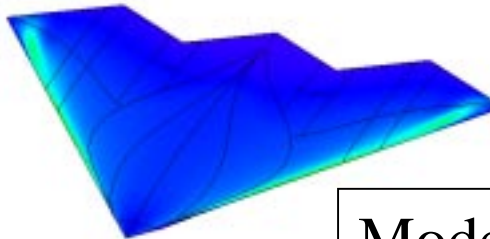
Flat Plate v Symmetric airfoil, CFD Euler, local vel contours

# EFFECT OF THICKNESS (FIN OFF) BAE SYSTEMS

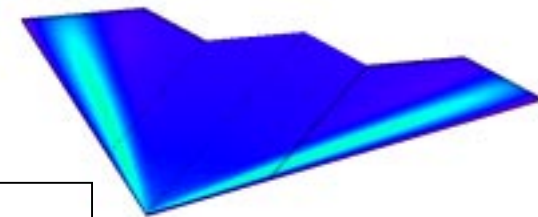
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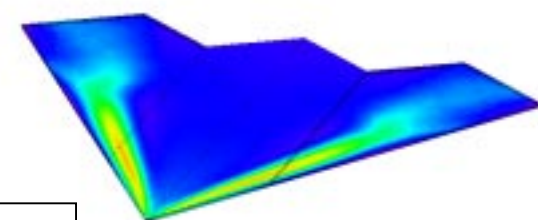
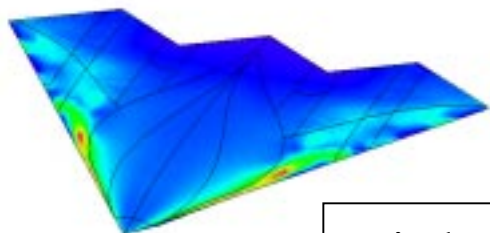
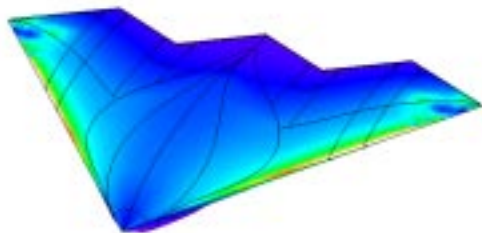
5% t/c



Flat Plate



Moderate  
incidence



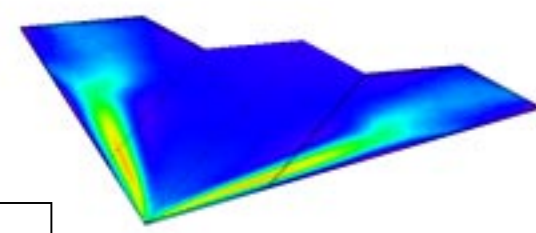
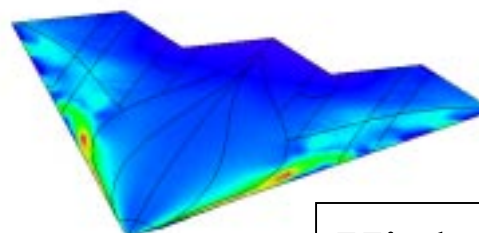
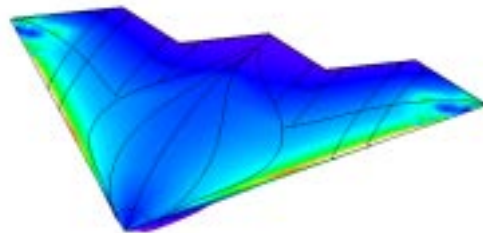
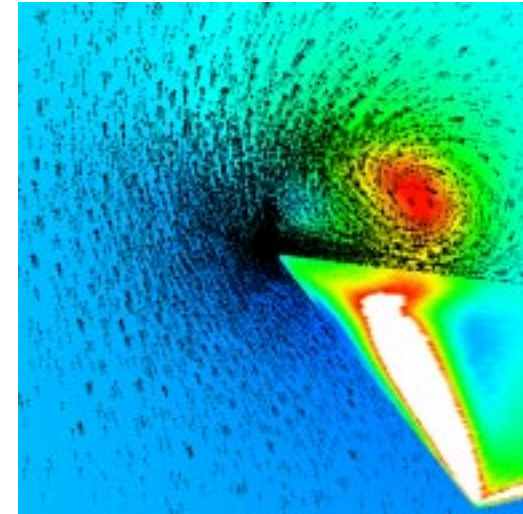
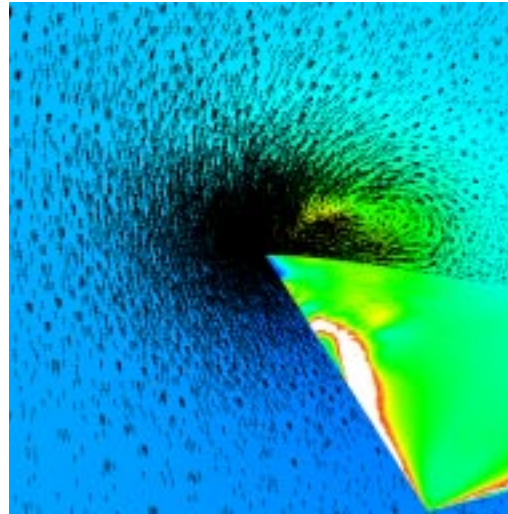
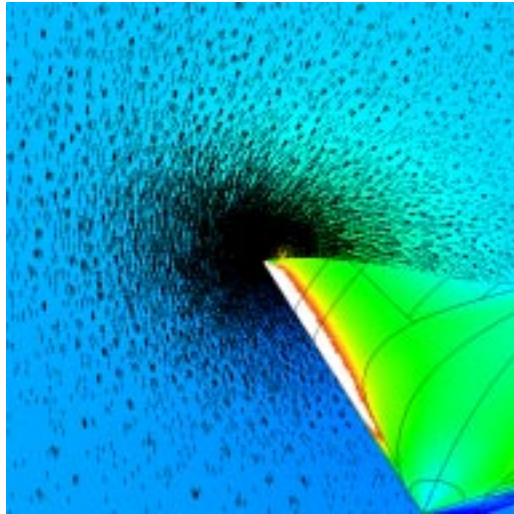
High  
Incidence

Flat Plate v Symmetric airfoil, CFD Euler, local vel contours



# EFFECT OF T/C ON FLOWFIELD

BAE SYSTEMS



High Inc

10% t/c

5% t/c

Flat plate

CFD Euler, velocity vectors and local vel contours

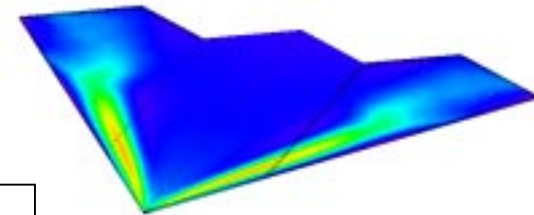
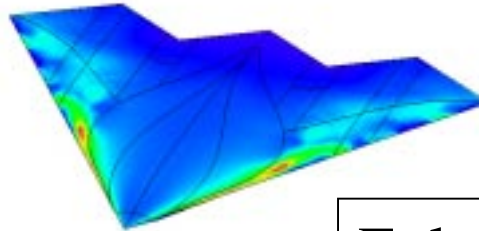
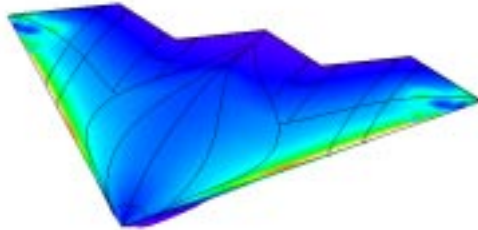
# EFFECT OF T/C, EULER v RANS

BAE SYSTEMS

10% t/c

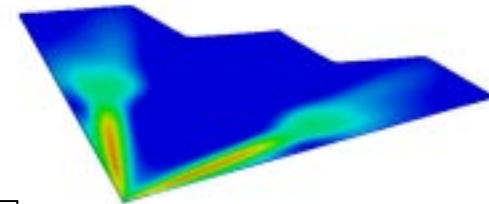
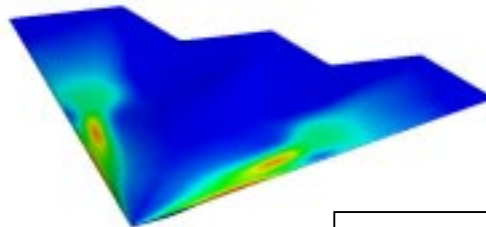
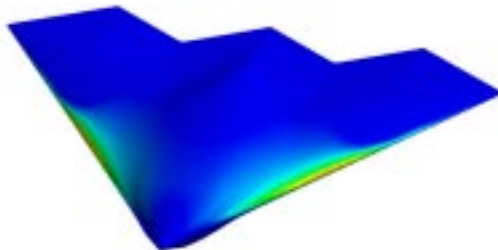
5% t/c

Flat Plate



Euler

High Inc



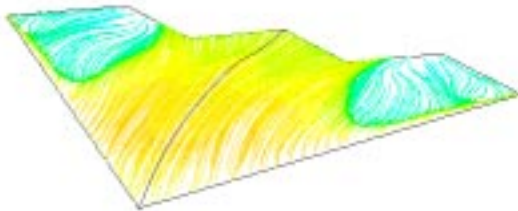
RANS

Flat Plate v Symmetric airfoil, Euler v RANS

# EFFECT OF THICKNESS

BAE SYSTEMS

10% t/c



5% t/c



Flat Plate



High Inc

RANS

CFD RANS, surface flow patterns

# SUMMARY

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- Euler showing good prediction of flat plate
- Absolute values of pitching moment poor at high incidence, though engineering decisions can be made by interpretation
- RANS improves absolute predictions, though at too great an overhead in CPU time to be practical for design optimisation
- Difference in flow behaviour between thin and thick airfoils defines limit of applicability of flat plate wind tunnel models



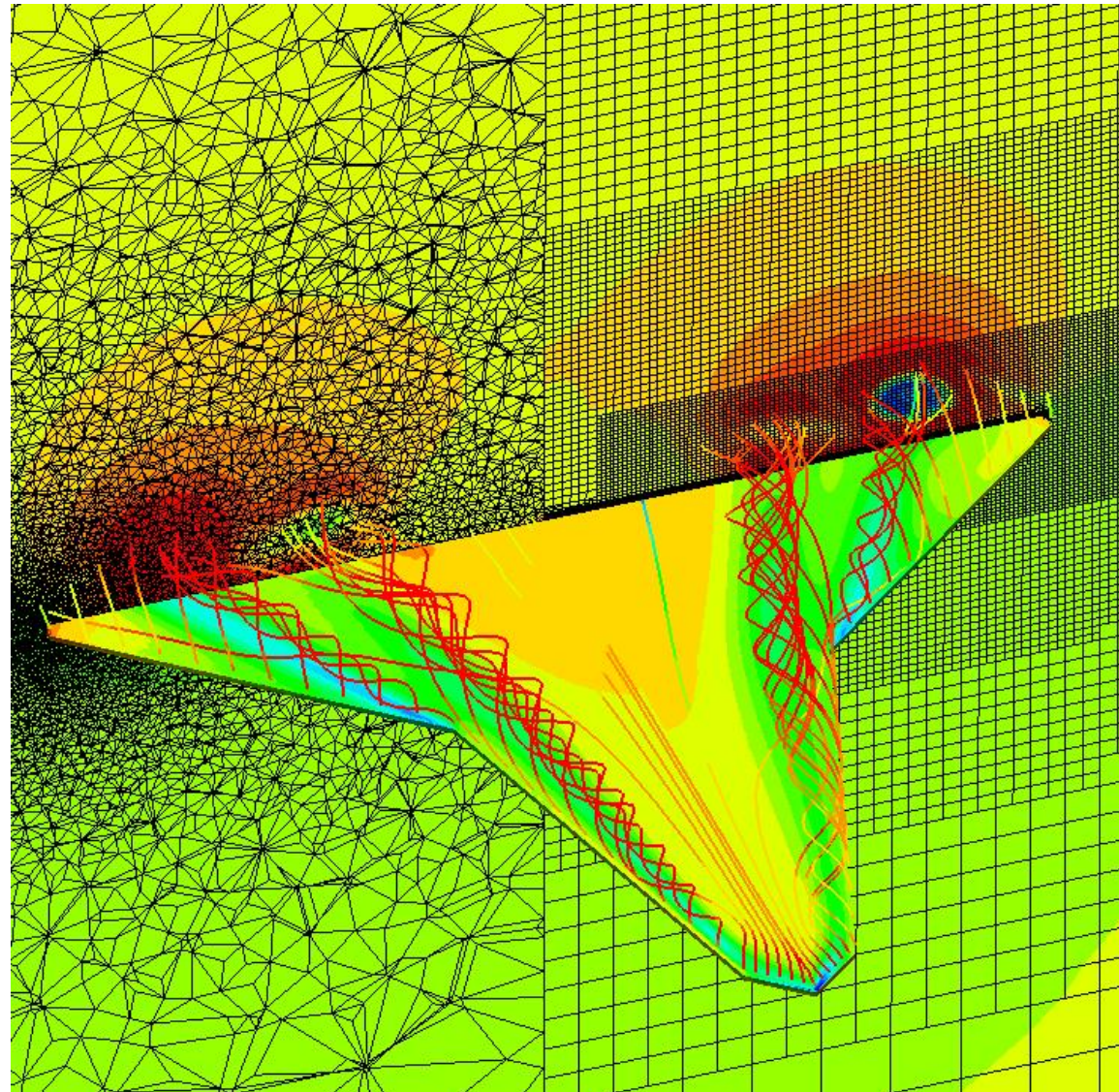
# FURTHER ASSESSMENT OF CFD

- RANS assessed on BAE Systems Autogrid meshes for a vortical flow case and a mixed attached/separated flow case
- kg results poor for both cases in terms of comparison with limited WT data, RANS ( $k\epsilon$  RNG) and engineering judgement
- $k\epsilon$  RNG results good for both cases

# VORTICAL FLOW CASE

BAE SYSTEMS

EULER

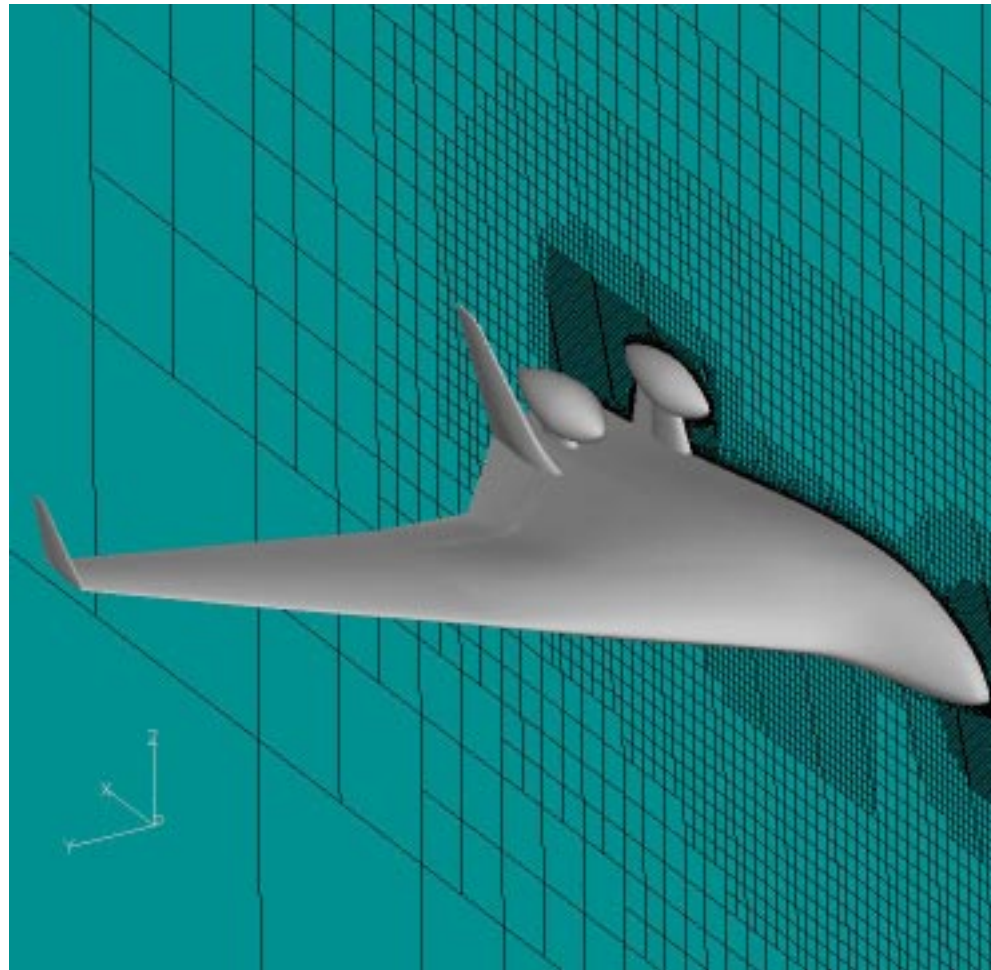


RANS

# MIXED ATTACHED/SEPARATED FLOW CASE

BAE SYSTEMS

BWB

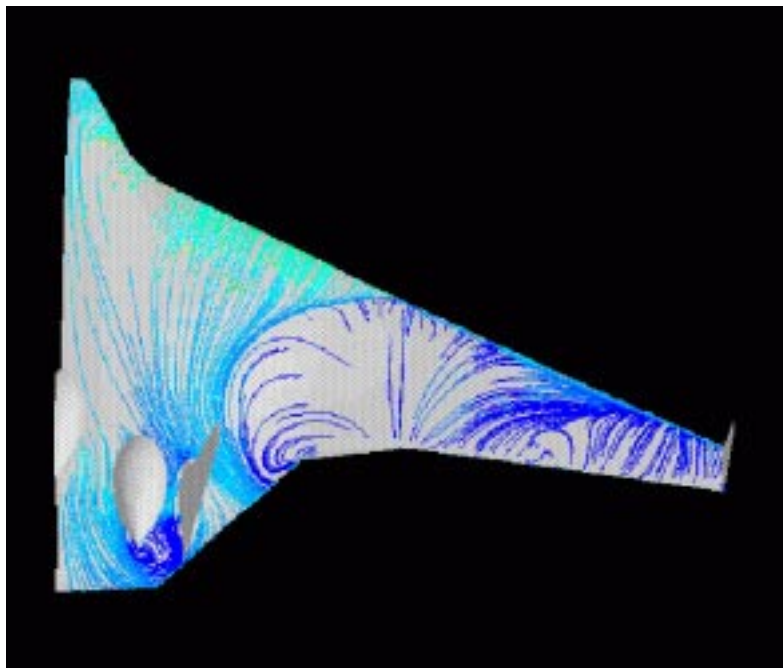




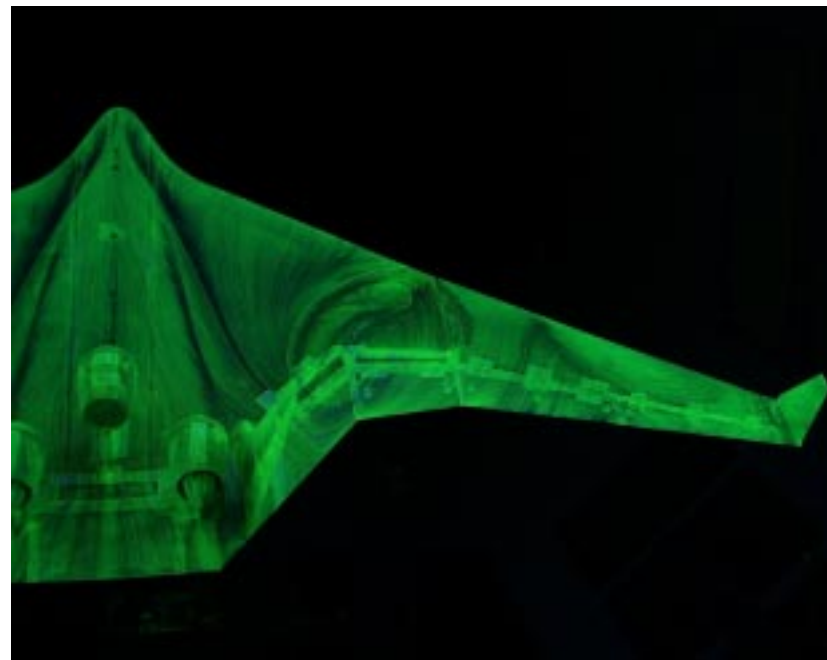
# MIXED ATTACHED/SEPARATED FLOW CASE

BAE SYSTEMS

BWB High Incidence



RANS KERNG



WIND TUNNEL

# CONCLUSIONS

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- Novel Planforms mean S+C Issues must be addressed early in the UCAV design cycle
- CFD and WT must work together here
- Requirement for rapid assessment
- Flat-plate and stereo-lith small-scale WT models, in conjunction with Euler and 'reduced-accuracy' RANS CFD can be applied here
- This approach requires engineering judgement and expertise to be fully effective